Amendments to the Claims:

1 – 10 (canceled)

11. (currently amended) An open-cooled blade~~component~~ for a gas turbine, comprising:

a root portion; and

an airfoil portion, wherein the airfoil portion comprises:

an outer wall exposed to a hot gas~~;~~,

a first cavity partly defined by the outer wall and for a first medium~~;~~,

a plurality of through-openings ~~are~~ arranged in the outer wall ~~and~~ where the through-openings open into the first cavity on a first side and into the hot-gas space on a second side~~;~~, and

a second cavity for admixing a second medium, the second cavity being fluidically connected to the through-openings,

wherein the second cavity is formed by supply passages that are provided in the outer wall and are connected via transverse passages to the through-openings designed as through-bores, so that the two media cannot be mixed until inside the through-bores.

12. (currently amended) The ~~component~~ blade as claimed in claim 11, wherein the outer wall has a multiplicity of through-bores, a multiplicity of supply passages running between the bores, and a multiplicity of further transverse passages linking the supply passages with the through-bores.

13. (currently amended) The ~~component~~ blade as claimed in claim 11, wherein the outer wall has at least two layers which can be connected to one another.

14. (currently amended) The ~~component~~ blade as claimed in claim 11, wherein the passages are incorporated between two layers in a layer surface.

15. (currently amended) The ~~component~~ blade as claimed in claim 11, wherein the first cavity is connected to a first fluid source and the supply passages can be connected to a second fluid source.

16. (currently amended) The ~~component~~ blade as claimed in claim 15, wherein one of the two fluid sources is an oxidation source and the other fluid source is a fuel source.

17. (canceled).

18. (currently amended) A combustion chamber for a gas turbine, comprising:

a ~~component designed as a~~ plurality of wall ~~element~~elements, ~~comprising;~~that form a combustion chamber, each element having:

an outer wall exposed to a hot gas~~;~~,

a first cavity partly defined by the outer wall and for a first medium~~;~~,

a plurality of through-openings ~~are~~ arranged in the outer wall ~~and~~ where the through-openings open into the first cavity on a first side and into the hot-gas space on a second side~~;~~, and

a second cavity for admixing a second medium, the second cavity being fluidically connected to the through-openings,

wherein the second cavity is formed by supply passages ~~that are~~ provided in the outer wall and ~~are~~ connected via transverse passages to the through-openings designed as through-bores, so that the two media cannot be mixed until inside the through-bores.

19. (currently amended) A gas turbine, comprising:

a compressor section;

a turbine section; ~~and~~

a combustion chamber~~,~~; ~~comprising~~ and

a plurality of blades where each blade comprises:

an outer wall exposed to a hot gas~~;~~,

a first cavity partly defined by the outer wall and for a first medium~~;~~,

a plurality of through-openings ~~are~~ arranged in the outer wall ~~and~~ where the through-openings open into the first cavity on a first side and into the hot-gas space on a second side~~;~~, and

a second cavity for admixing a second medium, the second cavity being fluidically connected to the through-openings,

wherein the second cavity is formed by supply passages ~~that are~~ provided in the outer wall and ~~are~~ connected via transverse passages to the through-openings designed as through-bores, so that the two media cannot be mixed until inside the through-bores.